Registration No.

26300



Quadrotor for Increased Situational Awareness for Ground Vehicles

Robert Severinghaus John Kaniarz

DISTRIBUTION STATEMENT A. Approved for public release; distribution is unlimited.

26 January 2015

U.S. Army Tank Automotive Research, Development, and Engineering Center Detroit Arsenal Warren, Michigan 48397-5000

Form Approved REPORT DOCUMENTATION PAGE OMB No. 0704-0188 Public reporting burden for this collection of information is estimated to average 1 hour per response, including the time for reviewing instructions, searching existing data sources, gathering and maintaining the data needed, and completing and reviewing this collection of information. Send comments regarding this burden estimate or any other aspect of this collection of information, including suggestions for reducing this burden to Department of Defense, Washington Headquarters Services, Directorate for Information Operations and Reports (0704-0188), 1215 Jefferson Davis Highway, Suite 1204, Arlington, VA 22202-4302. Respondents should be aware that notwithstanding any other provision of law, no person shall be subject to any penalty for failing to comply with a collection of information if it does not display a currently valid OMB control number. PLEASE DO NOT RETURN YOUR FORM TO THE ABOVE ADDRESS. 3. DATES COVERED (From - To) 1. REPORT DATE (DD-MM-YYYY) 2. REPORT TYPE 08-01-2015 **Technical** 03JAN2014 - 10NOV2014 4. TITLE AND SUBTITLE 5a. CONTRACT NUMBER **5b. GRANT NUMBER Quadrotor for Increased Situational Awareness for Ground Vehicles** 5c. PROGRAM ELEMENT NUMBER 6. AUTHOR(S) 5d. PROJECT NUMBER 5e. TASK NUMBER Robert Severinghaus 5f. WORK UNIT NUMBER John Kaniarz 7. PERFORMING ORGANIZATION NAME(S) AND ADDRESS(ES) 8. PERFORMING ORGANIZATION REPORT NUMBER **US Army TARDEC** 6501 E. 11 Mile Road 26300 Warren, MI 48397-5000 9. SPONSORING / MONITORING AGENCY NAME(S) AND ADDRESS(ES) 10. SPONSOR/MONITOR'S ACRONYM(S) 11. SPONSOR/MONITOR'S REPORT NUMBER(S) 12. DISTRIBUTION / AVAILABILITY STATEMENT A. Approved for public release; distribution is unlimited. 13. SUPPLEMENTARY NOTES 14. ABSTRACT This project received TARDEC Innovation Project funding in January 2014 to explore the concept of employing a quadrotor to improve situational awareness for ground vehicles in urban environments. This report details the modeling & simulation effort and the indoor flying quadrotor effort. 15. SUBJECT TERMS Innovation project, quadrotor, multirotor 16. SECURITY CLASSIFICATION OF: 17. LIMITATION 18. NUMBER 19a. NAME OF RESPONSIBLE **OF ABSTRACT OF PAGES PERSON** Robert Severinghaus a. REPORT c. THIS PAGE 19b. TELEPHONE NUMBER (include b. ABSTRACT none **UNCLAS UNCLAS UNCLAS** area code) 586-282-6722 15

Standard Form 298 (Rev. 8-98) Prescribed by ANSI Std. Z39.18

TABLE OF CONTENTS

1.0	INTRODUCTION AND OBJECTIVES	4
1.1		
2.0	MODELING AND SIMULATION	5
3.0	QUADROTOR TEST FLIGHTS	7
3.1	TEST PLATFORMS AND MODIFICATIONS	7
3.2	TEST PROCEDURE AND RESULTS	8
4.0	CONCLUSIONS	.11
APPE	NDIX A	12

1.0 INTRODUCTION AND OBJECTIVES

This project received TARDEC Innovation Project funding in January 2014 to explore a concept. The concept was to determine if a quadrotor could improve situational awareness for ground vehicles in urban environments. The vehicle commander must maintain situational awareness near the vehicle, and this may not be met by existing systems. Such systems include mast-mounted EO/IR, which can be limited by height, and UAS, which may not be available to the vehicle commander. Some ideas to investigate included what camera resolutions would be useful, where these cameras should be positioned, and whether a tethered quadrotor was feasible. Work concluded in October 2014.

1.1 IDENTIFICATION OF ISSUES AND APPROACH

The team began with brainstorming of the various technical issues to be solved in such a system. Figure 1 provides a summary of the major technical issues to be overcome. The team focused on three areas with limited scope.

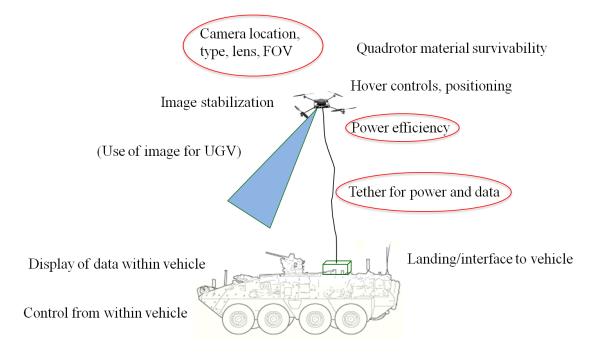


Figure 1. Technical issues for integration of quadrotor with vehicle.

Part of the effort was in modeling and simulation of camera location, type, lens, and field of view (FOV). With modification of the 360SA Visualization tool, the coverage of various configurations of cameras could be determined in a simulated urban environment.

The quadrotor flying part of the project was carried out by indoor flights. A temporary net structure and operating procedures provided an adequate risk reduction for these indoor flights. The flights focused on getting tether power for operation, and studying the efficiency of such a system.

2.0 MODELING AND SIMULATION

The surveillance performance of the system is affected by 3 key parameters: sensor resolution, FOV (lens zoom), and altitude. To measure this, we placed eight mannequins in a simulated urban environment using our in-house 360SA Visualization tool. To exhibit different effects we placed the mannequins in the following situations: on a 2nd story balcony; on a 3rd story roof partially obscured by parapets; in a courtyard behind a wall by a tree; on a sidewalk under an awning; in a courtyard behind a two story building, and three in plain view at 50m, 100m, and 200m.

We then selected various commercially available cameras and lenses. In all we measured all permutations of 3 standard camera bodies, 3 or 4 common lenses per camera, at six different altitudes. With each configuration we measured the number of camera pixels that "saw" each target (the "actual" count) as well as the number of pixels that could have seen the target had the view not been obstructed by buildings, trees, etc. (the "potential" count).

From the pixel totals we can calculate various measure of performance. The quantity of actual pixel hits is a measure the overall performance of the system detecting targets. The quantity of potential hits isolates the performance of the camera system due to range. The ratio of actual pixel hits to potential pixel hits isolates the situational performance of the system due to positioning.

Figure 2 shows a measurement from the tool. The intensity (whiteness) of a pixel increases with distance. The mannequin on the balcony has been colored red to highlight the actual pixels seen with bright red and the obstructed pixels with a darker red. The red pixels combined are the potential total.

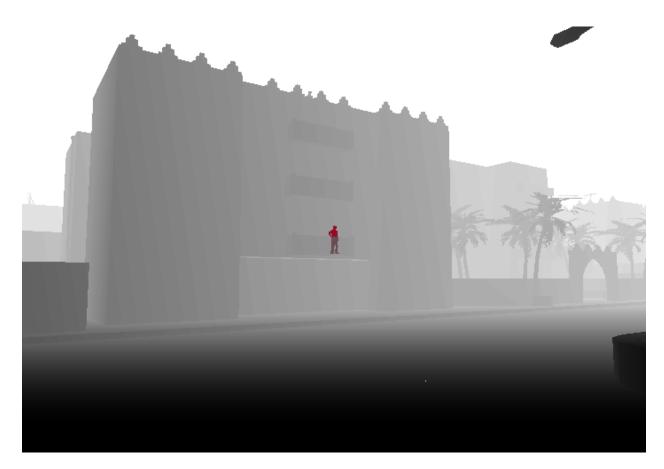


Figure 2. A measurement from the 360SA Visualization tool with target highlighted.

Looking at the resulting data in Appendix A we can see that the data properly exhibit basic properties of cameras. As a camera moves farther from a target the number of pixel hits decreases. Higher resolution cameras have more pixel hits on a given target. Wider angle lenses have fewer pixels on any one target as they view a larger area. Finally, from a given position, every configuration has the same ratio of potential to actual pixel hits on a given target (within rounding error) as this is a measurement of how obscured the target is.

Furthermore we can draw the following conclusions from the data. Increased resolution is always better – more pixels provide more detail. While the data shows that narrow lenses are better, this comes at the expense of situational awareness. Depending on the situation, particularly maximum range and size of targets of interest, there may not be one correct lens for the job. Until higher resolution sensor technology comes to market, an adjustable lens may be required.

One key takeaway regarding altitude is that there is no perfect altitude for the quadrotor. The target in the front courtyard exhibits this best. If the quadrotor is too low then it can't see the target over the wall. If it

flies too high then the target becomes obscured by a tree. Flying higher allows better vision behind the building but as the quadrotor flies higher the distance to the target increases thus making it harder to detect.

3.0 QUADROTOR TEST FLIGHTS

3.1 TEST PLATFORMS AND MODIFICATIONS

The flight testing used a commercial-off-the-shelf quadrotor, the 3D Robotics Quadrotor with Pixhawk autopilot, as shown in Figure 3.



Figure 3. Tethered quadrotor unpowered, on the floor within netted flying area.

The quadrotor (#3) was modified for tethered operation, as shown in Figure 4. The typical operating voltage of the quadrotor using a Lithium Polymer (LiPo) battery is 14.8V DC. This was replaced with a 48V to 15 V converter module, the Vicor V48A15C500BL. This was chosen for its size, reliability, and ability to provide 500W of power. Because the converter module was meant for board mounting, two printed circuit boards were fabricated to allow connection of the power converter to the tether and quadrotor battery connection, respectively. The converter module has a baseplate for thermal sink, which is on the bottom of the power converter. As it is mounted upside down, the baseplate is mounted about 1 inch from the bottom of the quadrotor, allowing airflow for heat dissipation.

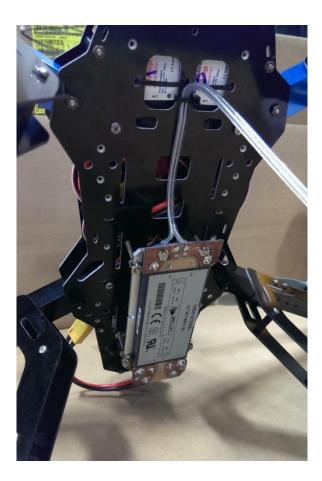


Figure 4. DC/DC converter mounted on bottom of quadrotor.

The quadrotor includes a power module that monitors voltage and current, and records these values at 1 Hz on the Pixhawk autopilot's micro-SD card. These logs provide a means to determine the total power used by the quadrotor, with 1 second resolution.

The base requires an AC-DC Converter that provides 48V DC at 500 W. The chosen power supply was a TDK-Lambda GWS50048. The voltage and current output of this power supply were monitored using a shunt.

The tether was 20 AWG, 2 wire, high voltage continuous flex flat cable. The insulation material was silicone rubber, allowing temperatures to 500F.

3.2 TEST PROCEDURE AND RESULTS

All operations of the quadrotor were conducted within the guideline of the SOP established for the TARDEC Indoor Quadrotor Laboratory. A separate risk management worksheet was completed and signed, outlining control measures for identified hazards. The flights were conducted within a net

suspended from the ceiling in the Ground Vehicle Robotics (GVR) high bay, with an approximate 12 foot cube of flying space, as in Figure 5. The net also covered the top of the cube. Although the quadrotor is capable of GPS flight, all flights were conducted indoors, without GPS lock. The quadrotor was then restricted to operation in stabilized or altitude hold modes. Most flights were radio controlled, in stabilized mode, requiring the near constant attention of the pilot.

A test was first performed with the original configuration quadrotor (#1) with a LiPo battery. These were relatively large 4000mAh 4S 35C batteries. From a starting voltage of 16.81V, the quadrotor flew for 11 minutes, 59 seconds to a final voltage of 14.47V. This quadrotor weighed 1374 grams, not including the battery weight of 417 grams.

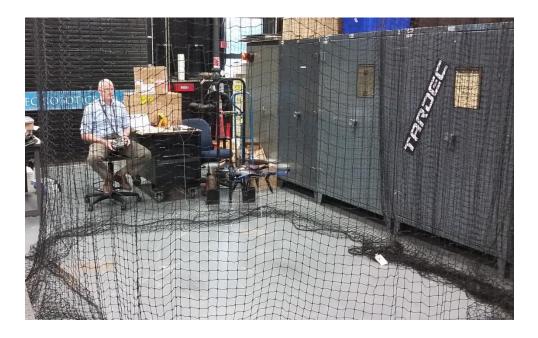


Figure 5. Tethered quadrotor in flight within netted flying area. Pilot at left.

A test was conducted to establish the operating characteristics of the tethered quadrotor (#3). First, the incremental power measurements were made with the tethered quadrotor on the ground, with results shown in Table 1. One value of note is the P=I²R value of power lost in the tether; even at 0.5 ohms of resistance in 20 feet of tether, the power lost at higher amperage is significant. It is at these higher amperage values that the DC/DC converter on the quadrotor becomes more efficient.

Table 1. Power values for tethered quadrotor on ground.

					Tether,				Dissipated	Efficiency	Fraction
		Power	Supply		20 ft.	Quadro	otor		by DC/DC	of DC/DC	to Quad
throttle											
setting		(V)	(A)	(W)	(W)	(V)	(A)	(W)	(W)	%	%
zero		47.92	0.39	18.7	0.1	14.95	0.07	1.0	17.6	5.6	5.6
prop spin		47.92	1.31	62.8	0.9	14.9	2.69	40.1	21.8	64.7	63.8
low	(on										
ground)		47.92	2.17	104.0	2.4	14.9	4.75	70.8	30.9	69.6	68.1
high	(on										
ground)		47.92	7.63	365.6	29.1	14.69	17.98	264.1	72.4	78.5	72.2
max	(on										
ground)		47.92	9.27	444.2	43.0	14.65	21.68	317.6	83.6	79.2	71.5

A test was conducted with a LiPo battery, and for a total quadrotor weight of 1.839 kg, it required 286 W in hover; which is 155 W/kg.

A test was conducted to determine the max thrust possible by the tethered quadrotor, this was performed by the configuration shown in Figure 6. A 2.26 kg weight was placed on the digital scale below the quadrotor, with a nearly taut lifting line. Max thrust was applied, but limited by the overcurrent protection of the AC-DC converter, at 11.1 Amps. For this test, the quadrotor lifted an additional 0.84 kg. It is possible that the quadrotor with conventional LiPo batteries could exceed this value.



Figure 6. Max thrust test setup for quadrotor, measuring AC-DC voltage, current, and the lifted weight value.

4.0 CONCLUSIONS

- 1. The safe voltage of 50V has excessive power losses in the tether for extended lengths. To avoid wasting this power, a higher DC transmission voltage should be chosen.
- 2. The tethered quadrotor can provide longer flight times than conventional LiPo battery powered quadrotors. There is a limit to the current provided at the ground, which can limit lifting power of the quadrotor.
- 3. When selecting a camera, higher resolution is always better. When selecting a lens, wider provides more situational awareness but narrow provides more detail on a target. With the resolutions available on currently available cameras a zoom lens is the only means to have it both ways.
- 4. There is no optimal altitude for the quadrotor. The height must be controllable to adjust for various situations. However, as performance decreases with range the quadrotor should be flown "high enough, but no higher".

APPENDIX A

U	877	222	110	51	159	144	51.48	81.2	1080	1920	50	1080p Medium
0	378	373	140	0	246	236	51.48			1920	40	1080p Medium
22	576	571	180	10	401	371	51.48	81.2	1080	1920	30	1080p Medium
121	801	732	231	210	625	489	51.48	81.2	1080	1920	20	1080p Medium
409	846	553	273	251	877	543	51.48	81.2	1080	1920	10	1080p Medium
148	567	154	271	0	753	321	51.48	81.2	1080	1920	0	1080p Medium
0	427	414	192	126	286	220	39.76	65.47	1080	1920	50	1080p Narrow
0	668	658	252	1	436	424	39.76	65.47	1080	1920	40	1080p Narrow
41	1041	1035	322	23	696	649	39.76	65.47	1080	1920	30	1080p Narrow
212	1415	1288	415	364	1125	879	39.76	65.47	1080	1920	20	1080p Narrow
710	1515	997	485	437	1566	972	39.76	65.47	1080	1920	10	1080p Narrow
958	1237	206	510	0	1671	672	39.76	65.47	1080	1920	0	1080p Narrow
0	11	8	6	4	8	4	87.92	104.25	480	704	50	ATSC Wide
0	19	19	10	0	11	9	87.92	104.25	480	704	40	ATSC Wide
2	29	29	11	0	20	17	87.92	104.25	480	704	30	ATSC Wide
7	41	38	14	14	32	25	87.92	104.25	480	704	20	ATSC Wide
23	42	29	15	15	43	28	87.92	104.25	480	704	10	ATSC Wide
29	35	6	15	0	44	17	87.92	104.25	480	704	0	ATSC Wide
0	27	25	13	8	20	14	65.47	81.2	480	704	50	ATSC Medium
0	43	43	17	0	28	26	65.47	81.2	480	704	40	ATSC Medium
3	73	71	21	1	43	40	65.47	81.2	480	704	30	ATSC Medium
12	99	90	25	25	82	64	65.47	81.2	480	704	20	ATSC Medium
45	106	71	32	32	111	67	65.47	81.2	480	704	10	ATSC Medium
70	88	14	33	0	119	49	65.47	81.2	480	704	0	ATSC Medium
0	52	48	23	13	31	21	51.48	65.47	480	704	50	ATSC Narrow
0	75	74	28	1	51	44	51.48	65.47	480	704	40	ATSC Narrow
5	131	130	43	2	88	79	51.48	65.47	480	704	30	ATSC Narrow
0	142	121	49	30	122	89	51.48	65.47	480	704	20	ATSC Narrow
90	184	124	61	61	190	119	51.48	65.47	480	704	10	ATSC Narrow
120	148	25	64	0	213	08	51.48	65.47	480	704	0	ATSC Narrow
Actual	Potential	Actual	Potential	Actual	Potential	Actual	VFOV	HFOV	VRes	HRes	Height	Camera
Overhang	Roof	Roof	Courtward	Colletvard	Ralconv	Ralconv						

17	17	39	39	88	86	120	119	134	50	1080p Medium
17	17	42	42	116	116	158	157	192	40	1080p Medium
16	16	50	50	158	158	221	219	271	30	1080p Medium
16	16	54	54	197	197	288	0	377	20	1080p Medium
16	16	54	54	232	232	351	0	496	10	1080p Medium
16	16	57	57	273	273	477	0	627	0	1080p Medium
25	25	77	77	158	155	209	203	239	50	1080p Narrow
26	26	85	85	205	205	285	282	334	40	1080p Narrow
26	26	90	90	278	278	392	391	483	30	1080p Narrow
26	26	96	96	355	355	517	0	666	20	1080p Narrow
26	26	100	100	418	418	631	0	864	10	1080p Narrow
26	26	101	101	438	438	686	0	958	0	1080p Narrow
0	0	0	0	6	4	7	6	8	50	ATSC Wide
0	0	0	0	6	6	10	10	10	40	ATSC Wide
0	0	0	0	10	10	13	13	14	30	ATSC Wide
0	0	0	0	16	16	16	0	20	20	ATSC Wide
0	0	0	0	16	16	21	0	28	10	ATSC Wide
0	0	0	0	17	17	18	0	29	0	ATSC Wide
0	0	3	2	13	12	15	14	15	50	ATSC Medium
0	0	5	5	18	18	18	18	22	40	ATSC Medium
0	0	4	4	22	22	26	25	31	30	ATSC Medium
0	0	5	5	26	26	36	0	45	20	ATSC Medium
0	0	6	6	28	28	45	0	57	10	ATSC Medium
0	0	5	5	29	29	50	0	71	0	ATSC Medium
0	0	12	12	19	19	23	23	26	50	ATSC Narrow
0	0	14	14	26	26	38	36	40	40	ATSC Narrow
0	0	12	12	32	31	49	48	60	30	ATSC Narrow
0	0	16	16	43	43	75	2	26	20	ATSC Narrow
0	0	18	18	51	51	76	0	110	10	ATSC Narrow
0	0	18	18	55	55	81	0	120	0	ATSC Narrow
Potential	Actual	Potential	Actual	Potential	Actual	Potential	Actual	Potential	Height	Camera
WalkFar	WalkFar	WalkMid	WalkMid	WalkNear	WalkNear	Behind	Behind	Overhang		

0	5671	5644	2562	1717	3747	3026	20.21	26.73	1944	2592	50	BlackFly f12.0
0	8923	8858	3386	5	5815	5779	20.21	26.73	1944	2592	40	BlackFly f12.0
543	13787	13678	4390	309	9370	8738	20.21	26.73	1944	2592	30	BlackFly f12.0
2876	18894	17193	5486	4662	14904	11700	20.21	26.73	1944	2592	20	BlackFly f12.0
9607	20223	13273	6428	5488	20914	13013	20.21	26.73	1944	2592	10	BlackFly f12.0
12960	16449	2694	6819	0	22328	9038	20.21	26.73	1944	2592	0	BlackFly f12.0
0	3184	3156	1448	964	2106	1690	26.73	35.16	1944	2592	50	BlackFly f9.0
0	4585	4556	1886	18	3163	3033	26.73	35.16	1944	2592	40	BlackFly f9.0
306	7746	7677	2473	178	5287	4930	26.73	35.16	1944	2592	30	BlackFly f9.0
0	8442	7415	2893	978	7088	5317	26.73	35.16	1944	2592	20	BlackFly f9.0
5390	11348	7450	3609	3106	11755	7319	26.73	35.16	1944	2592	10	BlackFly f9.0
7288	9274	1517	3835	0	12567	5085	26.73	35.16	1944	2592	0	BlackFly f9.0
0	982	968	442	293	651	514	46.31	59.39	1944	2592	50	BlackFly f5.0
0	1536	1523	582	1	1003	981	46.31	59.39	1944	2592	40	BlackFly f5.0
0	2073	2036	739	1	1496	1313	46.31	59.39	1944	2592	30	BlackFly f5.0
500	3282	2985	946	826	2588	2034	46.31	59.39	1944	2592	20	BlackFly f5.0
1665	3509	2305	1117	983	3631	2268	46.31	59.39	1944	2592	10	BlackFly f5.0
2250	2858	465	1190	0	3877	1566	46.31	59.39	1944	2592	0	BlackFly f5.0
0	432	418	194	128	287	221	65.89	81.65	1944	2592	50	BlackFly f3.3
0	670	660	255	1	437	425	65.89	81.65	1944	2592	40	BlackFly f3.3
40	1048	1041	326	21	707	657	65.89	81.65	1944	2592	30	BlackFly f3.3
216	1418	1290	420	370	1142	891	65.89	81.65	1944	2592	20	BlackFly f3.3
717	1530	1008	490	441	1582	980	65.89	81.65	1944	2592	10	BlackFly f3.3
969	1245	206	517	0	1691	679	65.89	81.65	1944	2592	0	BlackFly f3.3
0	108	100	49	33	71	49	71.75	104.25	1080	1920	50	1080p Wide
0	158	153	65	0	112	105	71.75	104.25	1080	1920	40	1080p Wide
10	262	259	82	7	176	161	71.75	104.25	1080	1920	30	1080p Wide
53	353	322	105	99	273	216	71.75	104.25	1080	1920	20	1080p Wide
176	386	251	122	120	390	241	71.75	104.25	1080	1920	10	1080p Wide
253	305	49	129	0	420	170	71.75	104.25	1080	1920	0	1080p Wide
Overhang Actual	Roof Potential	Roof Actual	Courtyard Potential	Courtyard Actual	Balcony Potential	Balcony Actual	VFOV	HFOV	VRes	HRes	Height	Camera
-	,	,	4		-							

1080p Wide 1080p Wide	10	253 217 169	0 0	163	110 107 93	110 107 93	25 26 26		1 1 1	1
1080p Wide	20	169	0	133	56	56	26	26	1	1
1080p Wide	30		100	1	72	72	26	26	1	1
1080p Wide	40		73	74	51	52	22	22	4 4	, ,
1080p Wide	50	969	51	689	442	442	102	102)5 25)5 25
BlackFly f3.3	10		0	636	422	422	101	101	26	26
BlackFly f3.3	20	674	0	517	356	356	97	97	26	26
BlackFly f3.3	30	487	394	395	283	283	96	96	26	26
BlackFly f3.3	40	336	282	286	206	206	85	85	26	26
BlackFly f3.3	50	240	207	210	156	158	77	77	25	25
BlackFly f5.0	0	2250	0	1586	1024	1024	232	232	52	52
BlackFly f5.0	10	2026	0	1463	965	965	231	231	53	53
BlackFly f5.0	20	1566	0	1191	826	826	224	224	52	52
BlackFly f5.0	30	1177	999	1008	687	688	215	215	54	54
BlackFly f5.0	40	771	656	665	493	493	192	192	51	51
BlackFly f5.0	50	540	475	481	365	365	174	174	49	49
BlackFly f9.0	0	7288	0	5126	3323	3323	737	737	175	175
BlackFly f9.0	10	6565	0	4746	3165	3166	742	742	174	174
BlackFly f9.0	20	5333	0	4459	2775	2775	732	732	180	180
BlackFly f9.0	30	3630	2925	2932	2100	2100	673	673	174	174
BlackFly f9.0	40	2586		2316	1660	1661	641	641	176	176
BlackFly f9.0	50	1772	1545	1555	1170	1170	564	564	170	170
BlackFly f12.0	0	12960	0	9108	5875	5875	1318	1318	305	305
BlackFly f12.0	10	11690	0	8432	5614	5614	1319	1319	315	315
BlackFly f12.0	20	9045	0	6882	4758	4758	1274	1274	312	312
BlackFly f12.0	30	6452	5187	5203	3736	3736	1199	1199	306	306
BlackFly f12.0	40	4483	3779	3816	2795	2796	1102	1102	302	302
BlackFly f12.0	50	3141	2757	2778	2092	2094	1006	1006	298	298